The Laser Interlaminar Reinforcement of Continuous Glass Fiber Composites

Interlaminar crack initiation and propagation are a major mode of failure in laminate fiber reinforced composites. A laser reinforcement process is developed to bond layers of glass fabric prior to the vacuum-assisted transfer molding of laminate composites. Glass fabric layers are bonded by fusing a dense glass bead to fibers within the laser focal volume, forming a 3D reinforcement architecture. Coupled heat transfer and viscous flow modeling is used to capture the temperature and morphology evolution of glass during the reinforcement process under experimentally observed conditions. Mode I double cantilever beam (DCB) testing is performed to quantify the effects of laser interlaminar reinforcements on composite delamination resistance. Postmortem high-resolution imaging of the fracture surface is used to characterize the toughening mechanism of the interlaminar reinforcements. Improved delamination resistance of laser reinforced composites derives from crack arrest and deflection mechanisms, showing a positive correlation to the reinforcement thickness. [DOI: 10.1115/1.4030754]

Keywords: glass fibers, delamination, mechanical testing, preform

1 Introduction

Laminate polymer matrix composite fabrication consists of the lay-up and consolidation of either pre-impregnated (prepreg) or stitched/woven fabric (preform) fiber reinforcements. Due to the prepreg lay-up process, all reinforcement fibers are aligned in the thickness direction. Prepreg laminates exhibit high fiber packing fractions and high strength along the fiber directions, but poor through thickness strength and fracture toughness. The insertion of reinforcing pins between layers of prepreg prior to curing, Z-pinning, has been shown to yield greater delamination resistance at the cost of planar strength [1]. Z-pins displace fibers in the lamina and introduce pockets of resin rich regions in an otherwise densely packed planar fiber architecture [2].

Fiber placement and resin infusion processing dictate the final form and reinforcement architecture of a composite part. Reinforcement architecture is the determining factor in composite mechanical properties. The ideal reinforcement architecture of a composite subject to multiaxial loads is often three-dimensional. This work develops a novel manufacturing process in the fabrication of preform laminate composites to improve interlaminar strength and to allow greater flexibility in their design and application.

Preform laminate fabrication benefits from lower handling costs (no need to refrigerate fabrics after matrix impregnation) and lower tooling costs (out of autoclave processing) over prepreg processing [3,4]. In conjunction with liquid resin transfer molding processes, stitched and woven fiber preforms with multidirectional fiber architectures are an attractive alternative to prepregs. Bilisik [5] and Bogdanovich [6] investigated weaving and stitching techniques to manufacture composites with three-dimensional fiber reinforcement preforms, which showed improved fracture toughness and through thickness strength. The introduction of woven or stitched fibers in the thickness direction displaces fibers in the planar directions similar to Z-pinning. Such 3D woven preforms exhibit greater out-of-plane strength and toughness but lower fiber volume fraction and in-plane strength than similar 2D laminates [7].

Laser joining of fibrous materials employs a direct fusion process, unlike mechanical or chemical bonding methods, which offers better seam quality and strength compared to stitching or weaving and has the potential to be applied in the fabrication of 3D reinforced preforms. The use of near infrared lasers to join thermoplastic fabrics has previously been explored for textile applications, such as air bags, medical fabrics, and protective garments, that require high-strength, high-precision, and seamless joints [8,9]. In these applications, the added cost of laser processing is justified by high weld rates, localized joints, watertight sealing, and reduced handling costs. Unlike stitching or weaving, laser processing of fabrics is a noncontact process and does not mechanically disrupt the nearby fiber architecture [10]. Laser interlaminar reinforcement offers the potential for high reinforcement density and direct fiber-to-fiber bonding in the Z direction, while allowing reinforcements to be restricted to localized regions, i.e., near existing stress concentrations.

As observed in textile joining investigations, a challenge in the fusion reinforcement of woven fibers is the undesirable formation of voids and discontinuities in the processing zone [8,9]. Woven textiles and composite preforms are composed of many fiber bundles, each consisting of hundreds of fibers. Flow and densification mechanisms induce local strains at the interface between the melt and the fiber preform because the relative density of the reinforcement is much greater than the initial material, as is common in sintering processes [11]. During laser fusion processing, densification and surface tension drive the melt to flow away from the center of the laser spot toward the low-density regions of the unprocessed fiber bundle, which forms voids within the process volume and leads to poor reinforcement strength. In order to achieve fully dense interlaminar reinforcements, we investigate the physical challenges posed by the densification and flow behaviors of the fiber preform during the fusion process.

One solution to the relative density and void formation problem in the laser joining of thermoplastic fabrics is to introduce a filler material into the process volume. These filler materials used consist of additive modified polymers that selectively absorb near infrared radiation (1064 nm), originally developed for the
transmission of E-glass fibers was achieved using an Nd:YAG laser I delamination.

To further characterize the crack propagation process during mode imaging of reinforcement and fiber fracture surfaces is performed

1. Interlaminar reinforcements bonded across four and eight glass ard D5528 for delamination resistance testing, as depicted in Fig.

Panels were cut into DCB specimens following the ASTM stand-

process using vacuum assisted resin transfer molding (VARTM).

reinforcement is formed by irradiating the fabric preform with a

within the fiber preform without a preferentially absorbing filler material. A two-phase, temperature-dependent flow model is used to simulate the melt pattern and reinforcement morphology observed during the laser process. The effects of laser interlaminar reinforcements on delamination resistance are characterized through mode I DCB delamination testing. Scanning electron microscopy (SEM) imaging of DCB fracture surfaces is used to investigate how laser interlaminar reinforcements affect the propagation of a delamination crack. The development of this novel interlaminar reinforcement method offers a new approach for fabricating 3D reinforcement architectures within a preform laminate composite.

2 Experimental Setup

Laser reinforcement of a plain woven E-glass fiber epoxy lami-

mate composite is performed during the fiber preform lay-up process prior to liquid resin infusion. A dense interlaminar reinforcement is formed by irradiating the fabric preform with a focused laser source and fusion bonding a soda-lime glass bead between the glass fabric layers. Laser processed glass fiber preforms are fabricated into laminate panels by a liquid resin infusion process using vacuum assisted resin transfer molding (VARTM). Panels were cut into DCB specimens following the ASTM standard D5528 for delamination resistance testing, as depicted in Fig. 1. Interlaminar reinforcements bonded across four and eight glass fabric layers were fabricated and tested using this method. SEM imaging of reinforcement and fiber fracture surfaces is performed to further characterize the crack propagation process during mode I delamination.

2.1 Through Thickness Laser Reinforcement. Laser fusion processing of E-glass fibers was achieved using an Nd:YAG laser from GSI Lumonics operating at 30 W. Glass reinforcements consist of 99.98% E-glass fibers with no surface coating prepared from 136 g/m² plain woven fabric with seven fiber bundles per centimeter and 300 fibers per bundle. Interlaminar reinforcements were formed by the introduction and cofusion of 1 mm diameter soda-lime glass beads between the glass fabric layers using a focused laser source. In order to limit the diameter of the fusion zone to the diameter of the fill bead, the laser was focused to a spot size of 0.8 mm using a final objective with a numerical aperture of 0.26. Fabric placement and motion control were accomplished using a Staubli six-axis robotic manipulator with a flat plate sample holder. A schematic for the laser irradiation of a glass fiber preform and the resulting melt morphology is depicted in Fig. 2(a).

Bulk E-glass and soda-lime glass are both highly transmitting at near infrared (1064 nm) radiation wavelengths. Without the addition of a preferentially absorbing filler material, an alternate means of laser energy absorption is needed to facilitate fusion bonding. This study found that densely packed E-glass fiber fabrics are strongly scattering and of sufficient opacity to absorb laser energy and melt the glass fibers. Investigations of radiation transmission through densely packed beds of glass spheres [12] and highly transmitting porous media [13] have similarly reported significant absorption due to multiple surface scattering. Multiple scattering-induced absorption of the laser is achieved through the increased mean free path of light traveling through the glass fibers. This mechanism is highly dependent upon the structure and packing density of the fiber preform, but was found to be controllable during laser processing.

Although other laser textile applications have evaluated the use of absorptive coatings and filler materials for targeted heating, it was determined from initial irradiation trials that surface coatings induce uneven absorption, which causes rapid uncontrollable vaporization in the sample and leads to retained porosity within the reinforcement. Packed beds of glass fibers with surfacing agents rapidly vaporize due to localized heating at the fiber surface and the dramatic increase in the absorption of infrared radiation at elevated glass temperatures [14]. The temperature of the sample becomes uncontrollable upon glass vaporization and plasma formation, resulting in the undesirable formation of voids within the focal volume. Laser heating by scattering-induced absorption is desired for the fusion processing of glass fibers, in order to avoid peak temperatures within the laser focal volume.

During laser processing, surface tension induced viscous flow is the dominant material transport mechanism for the fusion and densification of glass fiber preforms, as is consistent in the sintering of glass powders [15,16]. A major challenge to the fusion processing of fibrous systems is the high surface area and low

Fig. 1 (a) Sample schematic of a laser joined DCB specimen. (b) An image of a DCB test during displacement controlled load-

ing. Synchronized capture of high-resolution DCB fracture images enables the calculation of fracture energy with high spatial resolution.

Fig. 2 Interlaminar laser reinforcement process schematic showing (a) the irradiation of a stack of glass fabric resulting in the formation of a void in through the material with a dense ring of glass around the laser spot. (b) The through thickness laser reinforcement process employs a bead of dense glass fill to bond the glass melt, forming a dense reinforcement through the initial stack of glass fabrics.
2.2 Delamination Resistance Measurement. To measure the effect of laser interlaminar reinforcements on delamination resistance, DCB tests are conducted with reinforcements placed on the central axis of the beam at 10 mm intervals, as depicted in Fig. 1. After laser processing, fiber preforms are infused with System 2000 two part epoxy from FibreGlast industries using a VARTM method and cured for 24 hrs at room temperature. Laminate panels measuring 152 mm by 200 mm by 5 mm (32 ply) are cut and fabricated into DCB specimens as described by ASTM D5528. A 100 µm thick, 50 mm long polyethylene film is placed at the midplane of the DCB specimen 10 mm ahead of the interlaminar reinforcements to initiate the delamination crack during testing. Samples are tested under displacement control using an Instron 5948 material testing machine at a constant 5 mm/min cross head speed without unloading after crack initiation. Cross head extension is obtained from the Instron linear actuator and the load is measured by a 2 kN load cell with 0.01 N precision. A high-resolution synchronized camera (Point Grey GRAS-50S5M/C-C) with a 4x objective and data acquisition system was used to capture an image of the crack front at every 1 mm of cross head displacement while simultaneously recording the extension and load. Mode I fracture energy release rate is calculated using the modified compliance calibration method using the equation

\[ G_I = \frac{3P^2C_0^3}{2A_thh} \]  

where \( P \) is the load recorded, \( b \) is the beam width, \( h \) is the beam thickness, \( C = \delta/P \) is the compliance of the beam, and \( A_t \) is the slope of the least squares regression of the thickness normalized delamination length as a function of the cured root of compliance [18].

Load versus displacement and delamination resistance results obtained from the DCB tests quantify the fracture toughness along the crack plane. A 1 mm cross head displacement interval between data points yields sufficient spatial resolution to capture the significantly varying delamination resistance between processed and the nonprocessed regions along the crack front. Detailed spatial resolution from DCB tests coupled with postmortem imaging of the fracture surface gives added insight into the crack propagation process, allowing for the analysis of delamination propagation in the unreinforced laminate and the effect of crack arrest and deflection at the reinforcements. A group of eight DCB samples were tested for reinforcement thicknesses of four and eight laminae as well as a group of unreinforced laminates. The average delamination resistance, after an initial onset crack length (\( \Delta u > 5 \text{mm} \)), is calculated for each sample set.

3 Numerical Simulation

As laser energy transmits into the preform, fabric approaches the softening temperature of E-glass. Within the focus volume of the laser, significant temperature and viscosity gradients lead to the preferential flow along the fiber direction, which forms pockets of dense glass. A coupled two-dimensional heat transfer and compressible flow model is implemented to simulate the flow behavior of the molten pool during fiber processing. The model concurrently solves the compressible Navier–Stokes equations for viscous fluid flow, phase field (Cahn–Hilliard) equations for two-phase immiscibility and velocity-dependent heat equations, as discussed in detail in the previous paper [17]. The numerical model is validated using experimental results of the reinforcement morphology. Simulation results yield greater insight into the temperature-dependent behavior of fiber preforms than experiments alone, capturing the combined effects of laser heating, densification, and flow on the resulting reinforcement morphology.

Laser energy is assumed to be absorbed within a volume defined by a Gaussian distribution in the lateral and z directions. The laser power density is calculated from the total power input into the material over the laser interaction volume. As glass softens, the resistance to flow of the glass fiber preform and glass fiber melt is modeled by the temperature-dependent viscosity using the Vogel–Fulcher–Tammann (VFT) equation: given for E-glass and soda-lime glass as [19]

\[
\eta_{\text{E-glass}}(T) = 10^{-4.88} \cdot \frac{2850}{T-100}, \quad \eta_{\text{soda-lime}}(T) = 10^{-2.858} \cdot \frac{2850}{T-100}
\]  

where \( \eta \) is the glass dynamic viscosity (Pa·s) and \( T \) is the glass temperature in °C. The dynamic viscosity of E-glass decreases by eight orders of magnitude in the glass transition range between 700°C and 1400°C. Due to this highly temperature-dependent viscosity, glass behaves both as a solid and a liquid during the simulation. The dramatic reduction in glass viscosity dictates much of the physical response of the reinforcement morphology during reinforcement processing.

The glass fiber preform is modeled using the continuous media theory for selective laser sintering developed by Kolossov et al. [20], where the degree of fiber densification is defined by a sintering potential \( \pi \) given by

\[
\pi(x,t) = 1 - e^{-\int_0^{\zeta(T(x,t))} \eta(T) \, ds}
\]

where \( \pi \) is a continuous variable [0,1) corresponding to either separate fibers at \( \pi = 0 \) or fully dense glass at \( \pi = 1 \). The rate of densification \( \zeta(T) \) is first derived by Frenkel and later expanded upon [21–24] to be a temperature-dependent function given by

\[
\zeta = \frac{\gamma}{\eta(T) d_0}
\]

where \( \gamma \) is the surface energy, \( \eta(T) \) is the viscosity, and \( d_0 \) is a characteristic length scale of the initial material. Neither the surface tension (200 – 400 mJ/m²) nor the characteristic length of glass fiber (10 µm) is highly dependent upon temperature. The densification rate of glass fiber may thus be reduced to a function of temperature and viscosity

\[
\zeta(T) = K \frac{1}{\eta(T)}
\]

Viscosity and compaction rates of E-glass and soda-lime glass are plotted as a function of temperature in Fig. 3. The continuous media approximation of the glass fiber preform defined by Kolossov assumes a homogenous fractional density and thermal conductivity of partially dense glass given by

\[
\rho = (\alpha + (1-\alpha)\pi) \rho_{\text{bulk}}
\]

\[
k = (\alpha + (1-\alpha)\pi) k_{\text{bulk}}
\]

where \( \alpha = 0.4 \) defines the initial volume fraction of glass fibers and \( a = 0.1 \) defines the fractional conductivity between loose fibers and densely packed fibers to the bulk. Bulk thermal conductivity of glass is observed to be relatively constant with respect to temperature in the glass transition range [25]. At sufficient temperatures, glass spontaneously emits and absorbs infrared
radiation, leading to a temperature-dependent radiation factor in the thermal conductivity of the glass given by

\[ k_{rad}(T) = \frac{16\pi^2 \sigma T^3}{3 \tau} \]  

where \( n \) is the refractive index of the glass, \( \sigma \) is the Stefan–Boltzmann constant, and \( \tau \) is a geometric parameter relating to the absorption over the mean free path of the material [26,27]. Given the anisotropy of the fiber architecture in the planar directions, the material conductivity is assumed to be

\[ k_i = (a_i + (1-a_i)\pi)k + k_{rad} \]  

where the initial conductivity ratio in the \( a_i \) and \( a_f \) directions is assumed to be 0.1 and 0.01, respectively.

Anisotropic surface tension forces affect the flow of glass at the preform and melt interface caused by the fiber microstructure and glass density gradient at the melt boundary. A density gradient dependent body force is assumed to act on the melt volume given by

\[ F = \beta \nabla \rho V_{F,glass} \]

where \( F \) is the densification force and \( \beta \) is a constant tuning parameter used to match the morphology of the melt region during processing. The component of \( F \) in the thickness direction is zero given the anisotropic tendency of molten glass to flow along the fiber directions [17].

The model is bounded in a semi-infinite domain with a single horizontal parting line separating the initial volume fraction domains of air and glass fabric preform. Simulations of the laser fusion process include a bead of fully dense glass on the surface of the glass fiber preform. The initial condition of the sintering process include a bead of fully dense glass on the surface of the glass fiber preform. Melt flow is highly dependent upon the effect of multiple scattering on the mean free path taken by the beam through the glass fiber preform. Melt flow is highly dependent upon the surface energy and viscosity of the preform. Highly localized heating within the laser focus volume induces large temperature and viscosity gradients within the fibers, which causes preferential flow along the fiber directions. The resulting morphology of the melt is a function of the laser energy profile. As low-viscosity glass fibers compact and flow away from the center of the laser focus, the incoming laser energy penetrates deeper into the preform and is no longer absorbed in that fabric layer. At a sufficient melt diameter, a self-limiting state exists such that the total mass of glass is conserved during the simulation.

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<td>( \sigma )</td>
<td>N/m</td>
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4 Results and Discussion

Laser inter laminar reinforcement are formed by the cusion of a soda-lime glass bead at the laser focus volume within a glass fiber preform. The glass bead acts as a filler to overcome the densification and flow effects during localized heating of welded glass fabrics. Reinforcement morphology and penetration are measured experimentally, and a numerical model is implemented to study the physical effects of viscous densification during the laser process. Strength and fracture toughness enhancement of the laser processed composite is measured experimentally from double cantilever testing of woven fabric composites reinforced in the midplane. Delamination resistance is found to be dependent upon the reinforcement thickness.

4.1 Preform Densification. Laser irradiation of tightly woven glass fibers causes rapid consolidation of glass within the laser focal volume. Fibers approaching the softening temperature of E-glass shorten and agglomerate to its nearest neighbor, forming pockets of dense glass. Anisotropic shrinkage and wetting of the molten glass into the surrounding fiber materials cause the material at the center of the focal volume to flow outward. The resulting morphology of the melt is a ring of solid glass around the focal volume, as depicted in Figs. 2(a) and 4.

The melt formation and separation behavior is quantified from morphology measurements from optical microscopy images of laser irradiated fiber fabrics (Fig. 4). The melt diameter and melt depth are plotted with respect to time in Fig. 5. During the glass densification process, liquid glass flows and expands in both diameter and depth as a function of exposure time. The diameter of the melt pool increases rapidly and approaches a maximum of approximately 1 mm, slightly larger than the laser spot size. The depth of the melt pool increases linearly, with no apparent maximum within the range of exposure times evaluated in this study. The flow and separation behavior of the glass melt is a characteristic of the laser energy input and densification of the fiber substrate.

Laser energy absorption is highly dependent upon the effect of multiple scattering on the mean free path taken by the beam through the glass fiber preform. Melt flow is highly dependent upon the surface energy and viscosity of the preform. Highly localized heating within the laser focus volume induces large temperature and viscosity gradients within the fibers, which causes preferential flow along the fiber directions. The resulting morphology of the melt is a function of the laser energy profile. As low-viscosity glass fibers compact and flow away from the center of the laser focus, the incoming laser energy penetrates deeper into the preform and is no longer absorbed in that fabric layer. At a sufficient melt diameter, a self-limiting state exists such that the fiber surface tension balances the viscous flow resistance and a quasi-static equilibrium is reached in the radial direction.

As the molten glass flows away from the focal point, laser radiation passes deeper into the material and the process is repeated in each successive layer. Transmission of the laser energy into the preform occurs linearly in time during the melt separation process...
as is observed by the linear trend in melt depth as depicted in Fig. 5. As glass melts and separates in the layer above, the layer below is exposed to more laser radiation and begins to melt before the maximum diameter is reached in the layer above. Given a collimated laser beam with low dispersion, the melt morphology of the fabric is expected to be constant in diameter with respect to depth. A focused laser beam was used in this study with a numerical aperture of 0.26, causing the beam profile to diverge away from focus. Beam divergence reduces the beam intensity away from the focal plane, increasing the melt diameter and limiting the reinforcement thickness. This effect was not observed in the range of preform thicknesses evaluated.

Coupled laser heating, densification, and flow observed during experimental trials are captured in the transient multiphysics numerical simulation, with a Gaussian heat source input into a continuous substrate. As depicted in Fig. 6, the simulation domain is broken up into two halves with a midplane boundary between air and the partially dense glass preform. Edges of the reinforcement are set to a 90 deg contact angle with the domain boundary. The domain boundary is set to room temperature throughout the simulation and is allowed to have nonzero velocities at the boundaries with air. The glass sintering potential, density, and temperature are solved concurrently and resolved in space and time within the computation domain. The temperature and density distribution of the glass preform is depicted in snap shots of the simulation domain in Fig. 6. The effects of bundle densification and flow dynamics on the morphology of the melt pool during laser processing are plotted as a function of time in Fig. 7 together with the experimental results. The simulation results of the melt diameter and depth are generally in good agreement with experimental results. The discrepancy in diameter at the beginning is likely due to a delay in laser power deployment.

The temperature from the numerical simulation is observed to increase rapidly at the beginning (about 3500 °C/s) and overshoot slightly due to superheating, followed with small fluctuations caused by the glass fabric layer-by-layer before becoming steady as the laser absorption, heat accumulation, and flow effects reach a quasi-steady state as depicted in Fig. 7. The melt morphology is driven by the temperature field of the solution, as dictated by the temperature-dependent viscosity and sintering potential. For a stationary laser heat source, the temperature and diameter of the melt reach a maximum, as is confirmed in experiments. Morphology agreement between the simulation and experiments suggest that this model is well suited for studying the laser reinforcement process.

4.2 Reinforcement Morphology. A dense reinforcement between multiple fabric layers is formed when laser radiation is transmitted into a glass fiber preform through a solid fill material. Solid glass fill in the form of soda-lime glass beads serves two functions in the reinforcement process: to facilitate laser energy transmission into the fabric and to replace the fabric material as it flows away from the laser heat source. During the laser reinforcement process, the fiber preform absorbs focused laser energy and conducts heat back into the soda-lime bead. Optical microscopy images of laser interlaminar reinforcements show a dense transparent core of fill material surrounded by a dense shell of fused fiber, as depicted in Figs. 8 and 9. The interface between the fill fiber materials is visible under optical microscopy because the index of refraction of soda-lime glass (1.52) and E-glass (1.56) is sufficiently different, as shown in Fig. 9.

Similar to the irradiation of glass fiber preforms, laser absorption is caused by the multiple scattering of near infrared radiation in focal volume beneath the fill bead. Multiple scattering induced absorption of glass diminishes as fibers compact into a dense reinforcement. As fibers coalesce and the number of scattering surfaces is reduced, laser radiation is transmitted deeper into the preform. Thus, laser absorption during glass reinforcement processing is a self-limiting process. As the glass melt penetrates through all layers of fabric, the laser energy absorbed by the preform is reduced, and a quasi-steady-state condition is achieved when the effects of heat conduction, laser absorption, and viscous flow are balanced. The densification and flow of the fabric preform allow for the addition of glass fill into the process volume, bonding with the initially low volume fraction glass fibers. As the fill material penetrates deeper into the preform, laser energy is transmitted further into the fiber fabric. The final morphology of the reinforcement is limited by the initial geometry of the fill material.

A spherical soda-lime bead with a 1 mm diameter, as used in this study, is sufficient to form a continuous reinforcement of up to eight fabric layers in depth (approximately 1.2 mm). Laser processing of fewer than eight layers of fabric may be formed with this process, resulting in a slightly larger reinforcement diameter. Processing in excess of eight fabric layers showed incomplete melt penetration and limited fiber connectivity. Preform fiber packing density played an important role in the
penetration process, requiring mechanical pressure to be applied on the fabric in order to ensure fill penetration through multiple layers. Reinforcement cross sections (Fig. 9) show that the reinforcement diameter remains relatively constant throughout its thickness and fiber connectivity is preserved in each layer of the fabric.

Numerical simulations of the laser interlaminar reinforcement process are carried out with the same modeling scheme as the preform densification process. The simulation domain is split between air and glass fiber preform in the same manner as discussed in Sec. 4.1 with the exception of a 1 mm diameter bead of fully dense glass above the fibers. The laser source is approximated with a Gaussian heat input at the interface between the glass bead and the perform surface. The morphology and temperature evolution during the fill-induced laser reinforcement process are plotted in Figs. 10 and 11. The melt morphology obtained from simulations in Fig. 10 captures the reinforcement structure shown in Fig. 9. Similar to the preform densification behavior, the reinforcement penetration behaves linearly in time while the reinforcement diameter increases rapidly early in the process. The diameter of the reinforcement continues to increase throughout the reinforcement process, unlike the behavior of the fiber preform without the presence of the filler material.

Fig. 6 Numerical output of the glass densification process without filler material at 0.2 and 2.4 s. The substrate is shown to form a void in the center of the laser focus with the maximum temperature at the base of the laser penetration depth.

Fig. 7 Time-dependent melt morphology: temperature depth and diameter obtained from numerical simulation. The melt diameter and the depth are shown to follow the same trend as the experiments observed under optical microscopy.

Fig. 8 Optical micrograph of a through thickness reinforcement formed through four layers of glass fabric using a dense filler glass. Note that the center of the reinforcement is fully dense with little to no porosity, as observed from the greater transparency than the surrounding fabric.

Fig. 9 Cross section optical micrograph taken of a four layer through thickness joint cross section. Note that the joint is mostly dense with little to no porosity throughout its thickness. The soda lime fill glass is observable in optical images from the contrasting index of refraction from the E-glass fiber melt.
A maximum temperature of the melt, plotted in Fig. 11, is quickly reached in the laser reinforcement process. This state of thermal equilibrium is achieved during reinforcement processing due to the self-limiting laser absorption and densification behaviors of the glass fiber preform. The maximum temperature of the melt pool is consistent with the softening range of E-glass. Time-dependent fiber densification and laser absorption effects captured in this model are shown to yield good morphological agreement with experimental results. From the agreement between numerical and experimental results, it is determined that the dynamics captured by the numerical model are the controlling mechanisms in the laser reinforcement process.

4.3 Delamination Resistance. Evaluation of laser interlaminar reinforcement effects on mode I delamination resistance has been carried out using DCB bending tests of plain woven glass fiber composite samples. While delamination is observed to propagate smoothly along the midplane of unreinforced glass composite beams, reinforced samples exhibit observable start–stop crack propagation behavior during delamination testing with rapid crack propagation between each reinforcement and periods of crack arrest at the reinforcements. The characteristic start–stop crack propagation behavior is observed in the discontinuous load versus extension results shown in Fig. 12. Characteristic delamination resistance measurements of laminate samples with and without laser interlaminar reinforcements are plotted in Fig. 13. The discontinuous loading curve observed in laser reinforced samples differs dramatically from the relatively smooth fracture process observed in unreinforced samples, also plotted in Fig. 13.

During the DCB testing, crack arrest at length intervals consistent with the reinforcement spacing suggests that delamination is inhibited by the laser reinforcement. Dense glass reinforcements formed between fabric plies is of sufficient strength to resist the continuous propagation of the delamination crack along the fiber/matrix interface. During periods of crack arrest, the DCB is strained until sufficient stresses develop within the beam to propagate the crack around the reinforcement by fracturing the bonded fibers. As a result of the higher stresses stored in the reinforced DCB sample, crack deflection and branching across lamina are often observed at laser reinforcements. Evidence of crack deflection and fiber fracture at laser reinforcements is captured in SEM images of the reinforcement fracture surface presented in Sec. 4.4. Average fracture energies calculated from repeated DCB tests are plotted in Fig. 14. It is observed in the figure that there is a positive trend between reinforcement thickness and fracture energy. Because fibers are bonded to a continuous reinforcement
fraction from 0.59 to 0.37 [28]. The 2D plane weave glass fiber traditional fiberglass composites with a reduction in fiber volume 0.6–0.8 kJ/m². Mouritz et al. reported that orthogonal 3D woven cally exhibit interlaminar fracture toughness values of around reinforced composites. Unidirectional glass–fiber composites typi-

Lasers are used to reinforce samples in the current study. Laser reinforcement density is held constant in all the samples tested. Laser processed samples overall exhibit higher fracture energies than nonreinforced samples. Higher peak fracture energies observed at each reinforcement location, in Fig. 13, suggest that the average fracture energy of the sample is dependent upon the number of reinforcements in plane.

The increase in average delamination resistance observed in the laser reinforced samples is comparable with other mechanically reinforced composites. Unidirectional glass–fiber composites typically exhibit interlaminar fracture toughness values of around 0.6–0.8 kJ/m². Mouritz et al. reported that orthogonal 3D woven fabrics exhibit double the delamination resistance (1.4 kJ/m²) of traditional fiberglass composites with a reduction in fiber volume fraction from 0.59 to 0.37 [28]. The 2D plane weave glass fiber composite used in the current study exhibits higher baseline fracture toughness (0.9 kJ/m²) with a lower initial fiber volume fraction (0.42). Laser-reinforced samples with similar fiber volume fractions exhibit a 50% increase in average fracture toughness (about 1.3 kJ/m²) with minimal disruption to the global fiber architecture. While Koh et al. reported a 200–500% increase in delamination resistance when 0.5–4% Z-pins by volume are introduced in a unidirectional carbon fiber composite [29], a direct comparison between laser reinforcement and Z-pinning is not available due to the predominant application of Z-pins in carbon/graphite fiber prepregs rather than glass fiber preform composites. Further work is needed to evaluate the effect of planar reinforce-
density on delamination resistance using the current laser process. A more recent paper proposed an interlaminar toughening method [30].

4.4 Fracture Surface Morphology. Contrasting fracture surfaces between unreinforced and reinforced laminate fracture surfaces are depicted in Figs. 15 and 16. Delamination of the unreinforced glass preform composite takes place at the fiber/matrix interface, leaving an undulating surface profile defined by the woven fiber structure. In these regions, only the characteristic brittle fracture of the matrix material is observed with little to no fiber fracture. In contrast to the unreinforced regions, the fracture surface at laser reinforcements is characterized by significant fiber fracture and out-of-plane crack propagation. Figure 16 illustrates that crack deflection and fiber fracture effectively lead to positive and negative features on the opposing sample fracture surfaces. These features remain from intact interlaminar reinforcements after the significant crack deflection and fiber fracture around the reinforcement. The effect of crack deflection is observed in the branching of cracks to adjacent layers of fabric during DCB testing. SEM images of the reinforcement surface after testing show a high density of fibers fractured at the surface of the exposed reinforcements (Fig. 17). Fiber fracture and crack deflection are two key mechanisms for the increase in fracture toughness observed in laser reinforced samples.

As observed during testing, rapid delamination through unreinforced laminate regions tended to occur after elastic energy is released from a highly stressed region in the beam. Rapid crack propagation sometimes resulted in the failure of the next reinforce-
ment. Reinforcement failure is characterized by the brittle fracture of the solid glass with little to no fiber fracture or out-of-plane crack deflection. Reinforcement failure is more pronounced in the eight-layer reinforcement group due to the higher stresses.
5 Conclusions

A method for the laser reinforcement of glass fabric preforms has been developed to selectively bond glass fiber composites in the laminate thickness direction. The application of laser fusion processing has been evaluated on glass fabrics used in the VARTM of an epoxy polymer matrix composite. Melt and reinforcement formation dynamics have been simulated using a multiphysics numerical method to model the viscous flow, densification, and heating effects of glass fiber preforms during the laser fusion process. Numerical simulations of reinforcement diameter and depth have been validated with measurements of fabric melt morphology observed under optical microscopy.

Mechanical testing of laser reinforced samples has shown an increase in mode I fracture toughness as a function of the reinforcement thickness. Toughening effects of interlaminar reinforcements are observed in both load versus extension and delamination resistance curves. SEM images of the fracture surface show significant fiber fracture and crack deflection at the reinforcements, corresponding to the experimentally observed increase in local fracture toughness. Further testing is required to determine the effects of laser interlaminar reinforcements on the laminate’s planar mechanical properties.

Delamination resistance increases in proportion to the laser interlaminar reinforcement thickness up to eight laminae. Further work is necessary to evaluate ways to increase the maximum laser reinforcement thickness. The current process is limited by the geometry and insertion method of the soda-lime fill material. Fill geometries with higher aspect ratios may be evaluated as a means to form interlaminar reinforcements in excess of eight fabric layers.

A novel laser interlaminar reinforcement process has been shown to be a viable method for the fabrication of 3D reinforced preforms to enhance the delamination resistance of woven glass–fiber composites. The effects of laser fusion on fiber architecture and morphology have been examined through experiments and numerical simulations. Mode I fracture toughness findings show an increased delamination resistance as a function of reinforcement thickness and in-plane density. Possible efforts to further improve mode I fracture toughness with this method have been discussed. The application of laser interlaminar reinforcements in areas of high delamination stress concentrations presents a flexible and effective method to mitigate the onset of delamination in preform fabricated composites.

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References


